Research Article

A Three-Dimensional Range Migration Algorithm for Downward-Looking 3D-SAR with Single-Transmitting and Multiple-Receiving Linear Array Antennas

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This paper presents a three-dimensional (3D) range migration algorithm (RMA) suitable for downward-looking 3D-SAR with single-transmitting and multiple-receiving linear array antennas (STMR-LAA). As the round-trip range equation in 3D-SAR with STMR-LAA is a dual square root, the signal spectrum in 3D wavenumber domain contains nonlinear phase terms besides constant and linear phase terms. In this paper, the approximate expression of the signal spectrum is derived by expanding the implicit phase term to its Taylor series. Then the constant and nonlinear phase terms are calculated and compensated by multiplying the wavenumber filters. Finally, a 3D wavenumber mapping is proposed to make the signal evenly sampled in 3D wavenumber domain. Some simulating results are presented to validate the correctness of the analysis and the feasibility of the algorithm. In addition, the required accuracy on the platform position is analyzed at the end of the paper.

1. Introduction

Synthetic aperture radar (SAR) is a well-developed technique for remote sensing. It can provide two-dimensional (2D) high-resolution radar images along azimuth and slant-range directions. However, since 2D radar images are obtained by projecting the three-dimensional (3D) distributed targets onto the 2D plane, they usually suffer from geometric distortions, such as foreshortening, layover, and so forth. Furthermore, conventional SAR system generally works at the side-looking mode and exhibits strong shadowing caused by buildings, hills, and valleys, which may result in the information loss of the explored area [1].

To overcome the disadvantages mentioned above, the concept of 3D-SAR imaging with linear array antennas (LAA) was proposed [1-3]. 3D-SAR with LAA combines the theories of real and synthetic aperture and can provide

high-resolution radar images in three dimensions: azimuth, ground-range, and elevation. In addition, 3D-SAR with LAA can alleviate shadowing effects by looking downwards.

Due to its advanced performance, 3D-SAR with LAA has attracted wide attention. Two airborne downward-looking 3D-SAR with LAA systems, DRIVE [4–7] and ARTINO [8–16], are being developed at ONERA and FGAN-FHR, respectively. The LAA used in these two systems are bistatic configuration [3, 10], which means that some antenna elements (or single antenna element) are used for transmitting and the others are used for receiving. As the round-trip range equation in bistatic configuration is composed of dual square roots, the imaging turns out to be more difficult than that in monostatic configuration. Giret et al. [3] made some simplifications to the dual square roots with Fresnel approximation and proposed an algorithm derived from monostatic imaging algorithm. A shortcoming

of the algorithm is the range error caused by Fresnel approximation. Klare et al. [8, 9] introduced the concept of virtual antenna elements formed by the mean positions of every single transmitting and receiving antenna element to transform bistatic configuration into monostatic configuration and proposed a 3D imaging algorithm for ARTINO with beamforming operation in cross-track direction. However, the positions of virtual antenna elements depend on target locations, especially for the near-field targets.

Meanwhile, Reigber [17] proposed a 3D range migration algorithm (RMA) for SAR tomography imaging. Lopez-Sanchez and Fortuny [18] and Fortuny [19] developed a 3D RMA for 2D planar scanning aperture in the near-field zone of the target. Tan et al. [20] proposed a 3D RMA for SAR tomography imaging with digital spotlight in the elevation direction. All of these 3D RMAs are suitable for the monostatic configuration, where the signal spectrum in 3D wavenumber domain only contains constant and linear phase terms. However, due to the dual square root in the bistatic configuration, the signal spectrum in 3D wavenumber domain contains nonlinear phase terms besides constant and linear phase terms. Therefore, the 3D RMAs [17–20] mentioned above cannot be applied to the bistatic configuration directly.

In this paper, we propose a 3D RMA which can be applied to downward-looking 3D-SAR with singletransmitting and multiple-receiving linear array antennas (STMR-LAA). Firstly, we derive the approximate expression of the signal spectrum by expanding the implicit phase term to its Taylor series, where quartic and higher-order phase terms are neglected because the phase error caused by these phase terms contributes little to the analysis. Secondly, after the removal of constant phase term, quadratic and cubic phase terms caused by the bistatic configuration are calculated and compensated by multiplying a wavenumber filter, where some approximations are considered to improve computational efficiency. Finally, a 3D wavenumber mapping is proposed according to the expression of the residual phase terms, and an interpolation is performed to make the signal evenly sampled in 3D wavenumber domain.

The paper is organized as follows. The imaging geometry is described in Section 2. The signal properties are analyzed, and a 3D RMA for downward-looking 3D-SAR with STMR-LAA is derived in Section 3, where the removal of constant phase term, the compensation of nonlinear phase terms, and the mapping and interpolation in 3D wavenumber domain are highlighted. Simulation results are shown in Section 4. The required accuracy on the platform position to generate the 3D images is discussed in Section 5, and Section 6 gives conclusions of this paper.

We consider the simplified imaging scenario in a 3D spatial domain *OXYZ*, and the coordinate axes are defined in Figure 1. The platform is supposed to fly at the altitude *H*

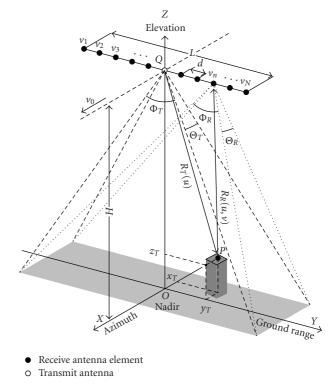


FIGURE 1: Imaging geometry of downward-looking 3D-SAR with STMR-LAA, where *x*-axis, *y*-axis and *z*-axis denote the azimuth (along-track), ground-range (cross-track), and elevation directions respectively, and *O* indicates the origin of the Cartesian coordinates.

along the *x*-axis with the constant velocity v_0 . The illuminated scenario is located at the nadir area of the platform, and the origin of the coordinate system is located at the nadir point.

The STMR-LAAs of the 3D-SAR are composed of a transmitting antenna element (Tx) and *N* receiving antenna elements (Rx) and distributed along the cross-track direction. Tx is located at the center of the LAA (i.e., *Q* in Figure 1), and its position is $\{u, 0, H\}$, where $u = v_0 t_m$, and t_m indicates the slow time. The entire Rxs are evenly spaced with distance *d*. Therefore, the length of LAA *L* equals to (N - 1)d, and the position of the *n*th Rx is $\{u, v, H\}$, where v = -L/2 + (n - 1)d and $n = 1, 2, 3, \ldots, N$.

Let a point target *P* locate at $\{x_T, y_T, z_T\}$ with reflectivity $\sigma(x_T, y_T, z_T)$. The distance from Tx to the target *P* is given by

$$R_T(u) = \sqrt{(u - x_T)^2 + y_T^2 + (H - z_T)^2}$$
(1)

shown in Figure 1, and the distance from *P* to the *n*th Rx is

$$R_R(u,v) = \sqrt{(u-x_T)^2 + (v-y_T)^2 + (H-z_T)^2}.$$
 (2)

Therefore, the round-trip range equation can be written as

$$R(u, v) = R_T(u) + R_R(u, v).$$
 (3)

3. Formulation of the 3D RMA

Here, we use the geometric relations and symbols illustrated in Figure 1 and assume that the transmitted signal p(t) is a chirp with the bandwidth *B*:

$$p(t) = \operatorname{rect}\left(\frac{t}{T}\right) \exp\left\{j2\pi\left(f_{c}t + \frac{1}{2}\mu t^{2}\right)\right\},\tag{4}$$

where *t* denotes the fast time; *T* indicates the pulse duration; f_c is the carrier frequency; μ is the chirp rate; the transmitted signal bandwidth $B = \mu T$; rect(\cdot) is a rectangular window function, which means rect(t/T) = 1 for $|t| \le T/2$, and zero otherwise.

3.1. Signal Spectrum in 3D Wavenumber Domain. After coherent demodulation, the received signal can be written as

$$s_{0}(t, u, v) = \iiint \sigma(x_{T}, y_{T}, z_{T}) \operatorname{rect}\left\{\left[t - \frac{R(u, v)}{c}\right]/T\right\}$$

$$\times \exp\left\{j\pi\mu\left(t - \frac{R(u, v)}{c}\right)^{2}\right\}$$
(5)
$$\times \exp\left\{-j2\pi f_{c}\frac{R(u, v)}{c}\right\}dx_{T}dy_{T}dz_{T},$$

where *c* is the speed of light. The antenna pattern and its type are not considered here since they contribute nothing to the essence of the analyzed themes.

The Fourier transform of the received signal $s_0(t, u, v)$ with respect to the fast-time *t* is

$$S_{0}(f_{t}, u, v) = \iiint \sigma(x_{T}, y_{T}, z_{T}) \operatorname{rect}\left(\frac{f_{t}}{\mu T}\right) \exp\left\{-j\pi \frac{f_{t}^{2}}{\mu}\right\}$$
$$\times \exp\left\{-j2\pi f_{t} \frac{R(u, v)}{c}\right\}$$
$$\times \exp\left\{-j2\pi f_{c} \frac{R(u, v)}{c}\right\} dx_{T} dy_{T} dz_{T},$$
(6)

where f_t is the slant-range frequency domain for t, and $|f_t| \le B/2$.

The slant-range compression is performed via the slant-range frequency domain matched filter:

$$H_1(f_t) = \operatorname{rect}\left(\frac{f_t}{\mu T}\right) \exp\left\{j\pi \frac{f_t^2}{\mu}\right\}.$$
 (7)

After slant-range compression, the signal can be written as

$$S_{1}(f_{t}, u, v) = S_{0}(f_{t}, u, v) \times H_{1}(f_{t})$$

$$= \iiint \sigma(x_{T}, y_{T}, z_{T})$$

$$\times \exp\left\{-j2\pi \frac{f_{t} + f_{c}}{c}R(u, v)\right\} dx_{T} dy_{T} dz_{T}.$$
(8)

Let $k = 2\pi (f_t + f_c)/c$ denote the wavenumber domain of the transmitted signal, and $k \in [k_{\min}, k_{\max}]$, where $k_{\min} = 2\pi (f_c - B/2)/c$, $k_{\max} = 2\pi (f_c + B/2)/c$ and $k_c = 2\pi f_c/c$; then (8) can be rewritten as

$$S_1(k, u, v) = \iiint \sigma(x_T, y_T, z_T) \exp\{-jkR(u, v)\} dx_T dy_T dz_T$$
(9)

in the slant-range wavenumber domain.

The 2D Fourier transform of the signal $S_1(k, u, v)$ with respect to (u, v) is

$$S_{2}(k, k_{u}, k_{v})$$

$$= \iint S_{1}(k, u, v) \exp\{-j(k_{u}u + k_{v}v)\} du dv$$

$$= \iiint \sigma(x_{T}, y_{T}, z_{T})$$

$$\times \left\{ \iint \exp\{-j[kR(u, v) + k_{u}u + k_{v}v]\} du dv \right\} dx_{T} dy_{T} dz_{T},$$
(10)

where k_u is the azimuth wavenumber domain for u, and k_v is the cross-track wavenumber domain for v. Let $U = u - x_T$ and $V = v - y_T$, then (10) can be rewritten as

$$S_{3}(k,k_{u},k_{v})$$

$$= \iiint \sigma(x_{T},y_{T},z_{T}) \exp\{-j(k_{u}x_{T}+k_{v}y_{T})\}$$

$$\times \left\{ \iint \exp\{-j[kR(U,V)+k_{u}U+k_{v}V]\} dUdV \right\} dx_{T}dy_{T}dz_{T},$$
(11)

where $R(U, V) = \sqrt{U^2 + y_T^2 + (H - z_T)^2} + \sqrt{U^2 + V^2 + (H - z_T)^2}$.

The principle of stationary phase (POSP) can be applied to solve the inner integral of (11). Here, we use $\Phi(U, V)$ to denote the negative phase term in (11), that is,

$$\Phi(U, V) = kR(U, V) + k_u U + k_v V.$$
(12)

Based on the POSP, the stationary points U_0 and V_0 should satisfy the equations set:

$$\frac{\partial \Phi(U, V)}{\partial U}\Big|_{U=U_0, V=V_0} = 0,$$

$$\frac{\partial \Phi(U, V)}{\partial V}\Big|_{U=U_0, V=V_0} = 0.$$
(13)

According to the definition of $\Phi(U, V)$ in (12), (13) can be written as

$$k \left[\frac{U_0}{\sqrt{U_0^2 + y_T^2 + (H - z_T)^2}} + \frac{U_0}{\sqrt{U_0^2 + V_0^2 + (H - z_T)^2}} \right] + k_u = 0,$$
(14)
$$k \frac{V_0}{\sqrt{U_0^2 + V_0^2 + (H - z_T)^2}} + k_v = 0.$$

It is too complicated to obtain the analytic solutions of equations set (14), and so we cannot have the explicit expressions of the stationary points U_0 and V_0 . However, both U_0 and V_0 are the function of k, k_u , k_v , y_T , and z_T , that is, $U_0 = U_0(k, k_u, k_v; y_T, z_T)$ and $V_0 = V_0(k, k_u, k_v; y_T, z_T)$. Therefore, the implicit expression of the signal spectrum in 3D wavenumber domain can be written as

$$S_{3}(k,k_{u},k_{v})$$

$$= \iiint \sigma(x_{T},y_{T},z_{T})$$

$$\times \exp\{-j(k_{u}x_{T}+k_{v}y_{T}+\Phi(U_{0},V_{0}))\}dx_{T}dy_{T}dz_{T},$$
(15)

where unessential multiplicative complex constants are not represented in the formula, and

$$\Phi(U_0, V_0) = kR(U_0, V_0) + k_u U_0 + k_v V_0.$$
(16)

 $\Phi(U_0, V_0)$ is written as Φ_0 for notational simplicity hereafter.

From (16), we can see that Φ_0 is the function of k, k_u , k_v , y_T , and z_T . If Φ_0 can be factorized to a combination of linear phase terms with respect to y_T and z_T , that is,

$$\Phi_0 = k_1(k, k_u, k_v) y_T + k_2(k, k_u, k_v) z_T,$$
(17)

then (15) can be written as

$$S_4(k_x, k_y, k_z)$$

$$= \iiint \sigma(x_T, y_T, z_T)$$

$$\times \exp\left\{-j\left(k_x x_T + k_y y_T + k_z z_T\right)\right\} dx_T dy_T dz_T,$$
(18)

where

$$k_x = k_u,$$

 $k_y = k_v + k_1(k, k_u, k_v),$ (19)
 $k_z = k_2(k, k_u, k_v).$

After the wavenumber interpolation and 3D IFFT operation, the 3D image in the spatial domain *OXYZ* can be generated. However, unlike the monostatic case [17–20], the Φ_0 in (16) contains nonlinear phase terms besides constant and linear phase ones. Since precise 3D RMA does not exist in 3D-SAR with STMR-LAA, only an approximate solution can be found.

In this paper, we expand Φ_0 to its Taylor series and compensate the phase terms except for the linear phase ones. The terms left in Φ_0 can be expressed as (17), and the following steps are the same as the monostatic case. Expanding Φ_0 to its Taylor series at $y_T = 0$, $z_T = 0$ and neglecting the quartic and higher-order terms, we have

$$\Phi_{0} \approx \Phi_{\text{const}} + \left(y_{T} \frac{\partial}{\partial y_{T}} + z_{T} \frac{\partial}{\partial z_{T}} \right) \Phi_{0} \bigg|_{y_{T}=0, z_{T}=0} + \frac{1}{2} \left(y_{T} \frac{\partial}{\partial y_{T}} + z_{T} \frac{\partial}{\partial z_{T}} \right)^{2} \Phi_{0} \bigg|_{y_{T}=0, z_{T}=0} + \frac{1}{6} \left(y_{T} \frac{\partial}{\partial y_{T}} + z_{T} \frac{\partial}{\partial z_{T}} \right)^{3} \Phi_{0} \bigg|_{y_{T}=0, z_{T}=0,}$$
(20)

where the first term is constant and independent of the target location; the second term is the linear phase term, which contributes to focusing; the third and forth terms are the main nonlinear phase terms, which will cause defocusing without compensation. The effect of every term in (20) will be analyzed in detail as follows.

3.2. Analysis and Removal of Constant Phase Term. The constant phase term Φ_{const} in (20) can be written as

$$\Phi_{\text{const}} = \Phi_0(k, k_u, k_v; \ y_T = 0, \ z_T = 0)$$

= $kR(U_{00}, V_{00}) + k_u U_{00} + k_v V_{00},$ (21)

where $U_{00} = U_0(k, k_u, k_v; y_T = 0, z_T = 0)$, $V_{00} = V_0(k, k_u, k_v; y_T = 0, z_T = 0)$, and

$$R(U_{00}, V_{00}) = \sqrt{U_{00}^2 + H^2} + \sqrt{U_{00}^2 + V_{00}^2 + H^2}.$$
 (22)

Since U_{00} and V_{00} are the special cases of U_0 and V_0 at $y_T = 0$, $z_T = 0$, they can be considered as the particular solutions of equations set (14). Thus, we have

$$k\left(\frac{U_{00}}{\sqrt{U_{00}^{2}+H^{2}}}+\frac{U_{00}}{\sqrt{U_{00}^{2}+V_{00}^{2}+H^{2}}}\right)+k_{u}=0,$$

$$k\frac{V_{00}}{\sqrt{U_{00}^{2}+V_{00}^{2}+H^{2}}}+k_{v}=0.$$
(23)

The analytic solutions of equations set (23) are

$$U_{00} = -\frac{k_{u}H}{\sqrt{\left(k + \sqrt{k^{2} - k_{v}^{2}}\right)^{2} - k_{u}^{2}}},$$

$$V_{00} = -\frac{k_{v}}{\sqrt{k^{2} - k_{v}^{2}}} \frac{\left(k + \sqrt{k^{2} - k_{v}^{2}}\right)H}{\sqrt{\left(k + \sqrt{k^{2} - k_{v}^{2}}\right)^{2} - k_{u}^{2}}}.$$
(24)

Thus Φ_{const} is equal to

$$\Phi_{\rm const} = H \sqrt{\left(k + \sqrt{k^2 - k_v^2}\right)^2 - k_u^2}.$$
 (25)

The removal of constant phase term can be performed by multiplying the following filter in 3D wavenumber domain:

$$H_2(k, k_u, k_v) = \exp\left\{jH\sqrt{\left(k + \sqrt{k^2 - k_v^2}\right)^2 - k_u^2}\right\}.$$
 (26)

3.3. Analysis and Compensation of Nonlinear Phase Terms. The nonlinear phase terms do not exist in the monostatic case but emerge here due to the bistatic configuration of LAA. Thus, the compensation of nonlinear phase terms is the key point of the 3D RMA proposed in this paper. According to (20), the nonlinear phase terms $\Phi_{nonline}$ in Φ_0 mainly include two parts, that is,

$$\Phi_{\text{nonline}} = \Phi_{\text{quad}} + \Phi_{\text{cubic}}, \qquad (27)$$

where Φ_{quad} denotes the quadratic phase term,

$$\begin{split} \Phi_{\text{quad}} &= \frac{1}{2} \left(y_T \frac{\partial}{\partial y_T} + z_T \frac{\partial}{\partial z_T} \right)^2 \Phi_0 \bigg|_{y_T = 0, \ z_T = 0} \\ &= \frac{1}{2} \left[\left. \frac{\partial^2 \Phi_0}{\partial y_T^2} \right|_{y_T = 0, \ z_T = 0} y_T^2 \\ &+ 2 \left. \frac{\partial^2 \Phi_0}{\partial y_T \partial z_T} \right|_{y_T = 0, \ z_T = 0} y_T z_T + \frac{\partial^2 \Phi_0}{\partial z_T^2} \bigg|_{y_T = 0, \ z_T = 0} z_T^2 \bigg|_{z_T}, \end{split}$$

$$(28)$$

and Φ_{cubic} denotes the cubic phase term,

$$\begin{split} \Phi_{\text{cubic}} &= \frac{1}{6} \left(y_T \frac{\partial}{\partial y_T} + z_T \frac{\partial}{\partial z_T} \right)^3 \Phi_0 \bigg|_{y_T = 0, \ z_T = 0} \\ &= \frac{1}{6} \left[\left. \frac{\partial^3 \Phi_0}{\partial y_T^3} \right|_{y_T = 0, \ z_T = 0} y_T^3 + 3 \left. \frac{\partial^3 \Phi_0}{\partial y_T^2 \partial z_T} \right|_{y_T = 0, \ z_T = 0} y_T^2 z_T \\ &+ 3 \left. \frac{\partial^3 \Phi_0}{\partial y_T \partial z_T^2} \right|_{y_T = 0, \ z_T = 0} y_T z_T^2 + \left. \frac{\partial^3 \Phi_0}{\partial z_T^3} \right|_{y_T = 0, \ z_T = 0} z_T^3 \right]. \end{split}$$

$$(29)$$

From (A.7) and (A.8) in Appendix A, the coefficients of Φ_{quad} associated with y_T^2 , $y_T z_T$, and z_T^2 in (28) can be, respectively, calculated as follows:

$$\begin{aligned} \frac{\partial^2 \Phi_0}{\partial y_T^2} \Big|_{y_T=0, z_T=0} \\ &= k \Biggl\{ \frac{1}{\sqrt{U_0^2 + y_T^2 + (H - z_T)^2}} - \frac{y_T^2}{\left[U_0^2 + y_T^2 + (H - z_T)^2\right]^{3/2}} \\ &- \frac{U_0 y_T}{\left[U_0^2 + y_T^2 + (H - z_T)^2\right]^{3/2}} \frac{\partial U_0}{\partial y_T} \Biggr\} \Bigg|_{y_T=0, z_T=0} \\ &= \frac{k \sqrt{\left(k + \sqrt{k^2 - k_v^2}\right)^2 - k_u^2}}{H\left(k + \sqrt{k^2 - k_v^2}\right)}, \end{aligned}$$
(30)

$$\begin{split} \frac{\partial^2 \Phi_0}{\partial y_T \partial z_T} \bigg|_{y_T = 0, \ z_T = 0} \\ &= k y_T \Biggl\{ \frac{H - z_T}{\left[U_0^2 + y_T^2 + (H - z_T)^2 \right]^{3/2}} \\ &- \frac{U_0}{\left[U_0^2 + y_T^2 + (H - z_T)^2 \right]^{3/2}} \frac{\partial U_0}{\partial z_T} \Biggr\} \bigg|_{y_T = 0, \ z_T = 0} \\ &= 0, \end{split}$$

$$\frac{\partial^2 \Phi_0}{\partial z_T^2} \Big|_{y_T=0, z_T=0} = \left\{ -\frac{k_u}{U_0} - \frac{k_u (H-z_T)}{U_0^2} \frac{\partial U_0}{\partial z_T} \right\} \Big|_{y_T=0, z_T=0} = 0,$$

$$= -\frac{k_u}{U_{00}} - \frac{k_u H}{U_{00}^2} \times \frac{\partial U_0}{\partial z_T} \Big|_{y_T=0, z_T=0} = 0,$$
(32)

where $\partial U_0/\partial z_T|_{y_T=0, z_T=0} = -U_{00}/H$ in (32) (see Appendix B). According to (30)–(32), there is only one term left in the quadratic phase term Φ_{quad} , that is,

$$\Phi_{\text{quad}} = \frac{k\sqrt{1-q}}{2H}y_T^2,\tag{33}$$

where

$$q = \frac{k_{\mu}^2}{\left(k + \sqrt{k^2 - k_{\nu}^2}\right)^2}.$$
 (34)

We can conclude that only the ground-range phase error term is included in Φ_{quad} , which will cause defocusing in the ground-range direction without compensation.

(31)

From (30)–(32), the coefficients of Φ_{cubic} associated with $y_T^3, z_T^3, y_T z_T^2$, and $y_T^2 z_T$ in (29) can be, respectively, calculated as follows:

$$\frac{\partial^3 \Phi_0}{\partial y_T^3} \bigg|_{y_T=0, z_T=0} = -\frac{2U_{00}}{\left(\sqrt{U_{00}^2 + H^2}\right)^3} \times \left. \frac{\partial U_0}{\partial y_T} \right|_{y_T=0, z_T=0} = 0,$$
(35)

$$\frac{\partial^3 \Phi_0}{\partial z_T^3} \bigg|_{y_T=0, \, z_T=0} = -\frac{k_u H}{U_{00}^2} \times \left. \frac{\partial^2 U_0}{\partial z_T^2} \right|_{y_T=0, \, z_T=0} = 0, \quad (36)$$

$$\frac{\partial^3 \Phi_0}{\partial y_T \partial z_T^2} \bigg|_{y_T=0, \ z_T=0} = 0, \tag{37}$$

 $\frac{\partial^3 \Phi_0}{\partial y_T^2 \partial z_T} \bigg|_{y_T=0, \, z_T=0}$

$$=\frac{k}{H\sqrt{U_{00}^{2}+H^{2}}}=\frac{k}{H^{2}}\sqrt{1-\frac{k_{u}^{2}}{\left(k+\sqrt{k^{2}-k_{v}^{2}}\right)^{2}}},\quad(38)$$

where $\partial U_0 / \partial y_T |_{y_T=0, z_T=0} = 0$ in (35) (see Appendix B), and $\partial^2 U_0 / \partial z_T^2 |_{y_T=0, z_T=0} = 0$ in (36) (see Appendix C). According to (35)–(38), there is only one term left in Φ_{cubic} , that is,

$$\Phi_{\text{cubic}} = \frac{k\sqrt{1-q}}{2H^2} y_T^2 z_T.$$
(39)

Thus, the nonlinear phase terms $\Phi_{nonline}$ in Φ_0 can be written as

$$\Phi_{\text{nonline}} = \frac{k\sqrt{1-q}}{2H^2} y_T^2 (H+z_T).$$
(40)

The compensation of Φ_{nonline} can be performed by multiplying the following filter in 3D wavenumber domain (i.e., (k, k_u, k_v) domain):

$$H_3(k, k_u, k_v; y_T, z_T) = \exp\{j\Phi_{\text{nonline}}\}.$$
 (41)

From (34), (40), and (41), we know that the compensation of Φ_{nonline} needs to take 5 variables into account, which are the slant-range wavenumber k, azimuth wavenumber k_u , cross-track wavenumber k_v , the location of target in the ground-range direction y_T , and elevation direction z_T . For the compensation of the nonlinear phase terms will require high computation time, some approximations will be considered to improve computational efficiency.

According to (9), the azimuth instantaneous wavenumber k_u is equal to

$$k_{u} = \frac{\partial}{\partial u} [-kR(u,v)]$$

$$= -k [\sin \theta_{T}(x_{T}, y_{T}, z_{T}) + \sin \theta_{Rn}(x_{T}, y_{T}, z_{T})].$$
(42)

In (42), $\theta_T(x_T, y_T, z_T) = \arcsin\{(u - x_T)/R_T(u)\}$, which is the azimuth instantaneous aspect angle of the Tx to the

point target *P*, and $|\theta_T(x_T, y_T, z_T)| \le \Theta_T/2$, where Θ_T is the azimuth beamwidth of the Tx. $\theta_{Rn}(x_T, y_T, z_T) = \arcsin\{(u - x_T)/R_R(u, v)\}$, which is the azimuth instantaneous aspect angle of the *n*th Rx to *P*, and $|\theta_{Rn}(x_T, y_T, z_T)| \le \Theta_R/2$, where Θ_R is the azimuth beamwidth of the Rx. Here, it is assumed that every Rx has the same azimuth beamwidth. Therefore, the maximum of k_u for a fixed slant-range wavenumber *k* is

$$k_{u\max} = k \left[\sin\left(\frac{\Theta_T}{2}\right) + \sin\left(\frac{\Theta_R}{2}\right) \right]. \tag{43}$$

Similarly, the cross-track instantaneous wavenumber k_v is equal to

$$k_{\nu} = \frac{\partial}{\partial \nu} [-kR(u,\nu)] = -k \sin \phi_{Rn}(x_T, y_T, z_T).$$
(44)

In (44), $\phi_{Rn}(x_T, y_T, z_T) = \arcsin\{(v - y_T)/R_R(u, v)\}$, which is the cross-track instantaneous aspect angle of the *n*th Rx to *P*, and $|\phi_{Rn}(x_T, y_T, z_T)| \le \Phi_R/2$, where Φ_R is the cross-track beamwidth of the Rx and every Rx is assumed to have the same cross-track beamwidth. The maximum of k_v for a fixed slant-range wavenumber *k* is

$$k_{\nu \max} = k \sin\left(\frac{\Phi_R}{2}\right). \tag{45}$$

According to (43) and (45), the maximum of q is

$$q_{\max} = \frac{k_{u\max}^2}{\left(k + \sqrt{k^2 - k_{v\max}^2}\right)^2} = \left[\frac{\sin(\Theta_T/2) + \sin(\Theta_R/2)}{1 + \cos(\Phi_R/2)}\right]^2.$$
(46)

If the parameters of 3D-SAR with STMR-LAA are chosen to satisfy $q_{\text{max}} \ll 1$, then the wavenumber filter shown in (41) can be approximated as

$$H_3(k; y_T, z_T) \approx \exp\left\{j\frac{k}{2H^2}y_T^2(H+z_T)\right\}.$$
 (47)

After approximation, the wavenumber filter only depends on k, y_T , and z_T , and the computational efficiency is improved. The nonlinear phase compensation can be performed by multiplying (47) in the slant-range wavenumber domain before the 2D Fourier transform with respect to (u, v) (i.e., (k, u, v) domain). The phase error between (41) and (47) will be discussed in Section 4.

3.4. Mapping and Interpolation in 3D Wavenumber Domain. After the removal of constant phase term Φ_{const} and compensation of nonlinear phase terms Φ_{nonline} , the residual phase term in Φ_0 can be written as

$$\Phi_{\text{line}} = \left(y_T \frac{\partial}{\partial y_T} + z_T \frac{\partial}{\partial z_T} \right) \Phi_0 \Big|_{y_T = 0, \ z_T = 0}$$

$$= \left. \frac{\partial \Phi_0}{\partial y_T} \right|_{y_T = 0, \ z_T = 0} y_T + \left. \frac{\partial \Phi_0}{\partial z_T} \right|_{y_T = 0, \ z_T = 0} z_T \qquad (48)$$

$$= -z_T \sqrt{\left(k + \sqrt{k^2 - k_v^2} \right)^2 - k_u^2},$$

where the coefficients of the phase terms with respect to y_T and z_T are derived in Appendix A. According to (48), the residual phase term in Φ_0 is composed of the linear phase terms with respect to y_T and z_T , and we can obtain that $k_1(k, k_u, k_v) = 0$ and $k_2(k, k_u, k_v) = -\sqrt{\left(k + \sqrt{k^2 - k_v^2}\right)^2 - k_u^2}$ by comparing (48) with (17). Thus we can define a 3D wavenumber mapping:

$$k_x = k_u, \tag{49a}$$

$$k_{\nu} = k_{\nu}, \tag{49b}$$

$$k_{z} = -\sqrt{\left(k + \sqrt{k^{2} - k_{v}^{2}}\right)^{2} - k_{u}^{2}},$$
(49c)

which is used to transform the signal in the (k, k_u, k_v) domain into the one in the (k_x, k_y, k_z) domain. The 3D wavenumber mapping used here is distinguished from the one used in the monostatic case [17–20]. It essentially replaces wavenumber in the direction of wave propagation by wavenumber observed in the azimuth, ground-range, and elevation directions. The reason for the negative sign in (49c) is that the defined direction between wave propagation and the elevation is opposite in this paper.

After the 3D wavenumber mapping shown in ((49a), (49b), (49c)), the available discrete evenly spaced samples of (t, u, v) are transformed into evenly spaced samples of (k_x, k_y) and unevenly spaced samples of the elevation wavenumber k_z . Thus an interpolation shown in (49c) is performed to transform these unevenly spaced samples into evenly spaced ones in the k_z domain.

Once the interpolation is done, we will obtain the expression shown in (18). After the 3D inverse Fourier transform of the signal $S_4(k_x, k_y, k_z)$ with respect to (k_x, k_y, k_z) , the reconstructed target image in a 3D spatial domain *OXYZ* can be expressed as

$$s_4(x, y, z) = \iiint \sigma(x_T, y_T, z_T) \operatorname{sinc} \left\{ \frac{B_{x_T}(x - x_T)}{2\pi} \right\}$$

$$\times \operatorname{sinc} \left\{ \frac{B_{y_T}(y - y_T)}{2\pi} \right\}$$

$$\times \operatorname{sinc} \left\{ \frac{B_{z_T}(z - z_T)}{2\pi} \right\} dx_T dy_T dz_T$$
(50)

 2π

where the sinc function is defined as
$$\operatorname{sinc}(a) = \frac{\sin(\pi a)}{(\pi a)}$$

 B_{x_T} is the bandwidth of the target *P* spectral support in the
 k_x domain, which determines the azimuth resolution; B_{y_T}
is the bandwidth of the target *P* spectral support in the k_y
domain, which determines the ground-range resolution; B_{z_T}
is the bandwidth of the target *P* spectral support in the k_z
domain, which determines the ground-range resolution; B_{z_T}
is the bandwidth of the target *P* spectral support in the k_z
domain, which determines the elevation resolution.

3.5. Summary of 3D RMA. So far, the 3D RMA for 3D-SAR with STMR-LAA has been deduced in 3D wavenumber domain. The block scheme of this algorithm is shown in Figure 2. The execution steps and some implementation details are explained as follows.

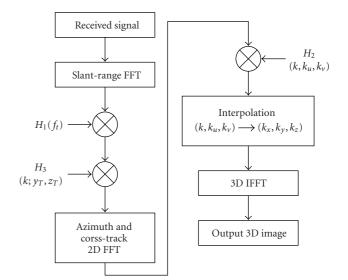


FIGURE 2: Block scheme of 3D RMA for 3D-SAR with STMR-LAA.

- (a) Fourier transform along the slant-range direction is firstly applied to the received signal, and the slant-range compression is performed with the frequency domain matched filter $H_1(f_t)$.
- (b) The nonlinear phase terms are compensated by multiplying the wavenumber domain filter $H_3(k; y_T, z_T)$, where some approximations are adopted in this step to improve the computational efficiency.
- (c) After the 2D Fourier transform along the azimuth and cross-track directions, constant phase term is removed by multiplying $H_2(k, k_u, k_v)$ in 3D wavenumber domain.
- (d) A 3D wavenumber mapping is proposed to transform the signal in the (k, k_u, k_v) domain into the one in the (k_x, k_y, k_z) domain, and an interpolation is done to make the signal evenly sampled in the (k_x, k_y, k_z) domain. The interpolation can be separated into two steps, where the first step is $k_3 = k + \sqrt{k^2 - k_v^2}$ and the following one is $k_z = -\sqrt{k_3^2 - k_u^2}$.
- (e) 3D inverse Fourier transform is performed to generate the final 3D image in the spatial domain *OXYZ*.

4. Simulation Results

For evaluating the proposed algorithm, we carry out some simulations for 3D-SAR with STMR-LAA in the downwardlooking mode. Simulation parameters are given in Table 1. The values of the platform parameters correspond to the characteristics of a small UAV.

Consider a $50 \text{ m} \times 50 \text{ m} \times 50 \text{ m}$ (azimuth \times ground – range \times elevation) target region composed of 28 point targets that are located at the nadir area under the platform in 3D spatial domain *OXYZ*. The distribution of point targets is shown in Figure 3.

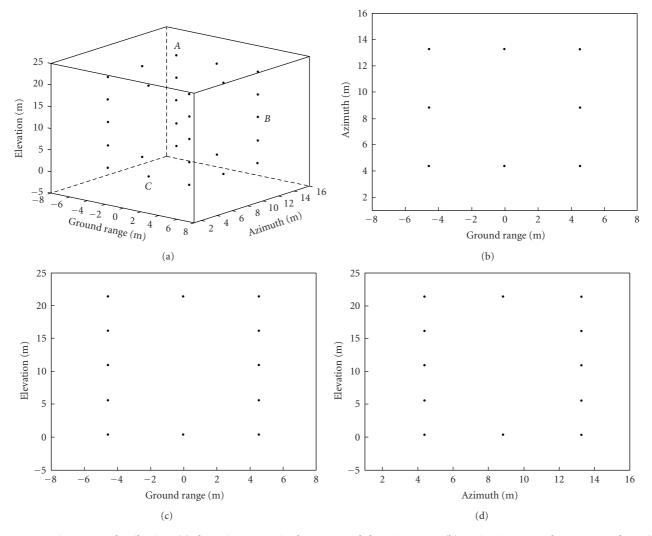


FIGURE 3: Point targets distribution. (a) the point targets in the 3D spatial domain *OXYZ*; (b) projections onto the 2D *OXY* plane; (c) projections onto the 2D *OYZ* plane; (d) projections onto the 2D *OXZ* plane.

According to (46) and simulation parameters in Table 1, we can obtain that $q_{\text{max}} = 1.9 \times 10^{-5}$. Figure 4 shows the nonlinear phase terms shown in (47) and the nonlinear phase error between $H_3(k, k_u, k_v; y_T, z_T)$ in (41) and $H_3(k; y_T, z_T)$ in (47), where y_T and z_T are chosen at the borders of the illuminated scenario. It can be seen that nonlinear phase errors are all smaller than $\pi/4$ radian at the three wavenumber points k_{\min} , k_c , and k_{\max} and can be negligible compared with the corresponding nonlinear phase terms. Therefore, it will not deteriorate the focusing quality by the use of the approximate expression (47) to compensate the nonlinear phase terms.

With the 3D RMA proposed in this paper, the 3D imaging results of the 28 point targets are shown in Figure 5, where an oversampling factor of 8 has been used in 5(b)–5(d) to enhance the intensity of the image display and no windowing is used in the processing. The 3D image is displayed at the threshold of -3 dB as shown in Figure 5(a), and the positions of the 28 point targets are consistent with the real situations in Figure 3(a). Figures 5(b) and 5(c) show

TABLE 1: Simulation parameters.

Altitude (<i>H</i>)	1 km
Velocity (v_0)	40 m/s
Carrier frequency (f_c)	37.5 GHz
Bandwidth (B)	300 MHz
Pulse repetition frequency	200 Hz
Length of LAA (L)	16 m
Number of $Rx(N)$	256
Azimuth beamwidth of Tx and Rx (Θ_T and Θ_R)	0.5°
Cross-track beamwidth of Rx (Φ_R)	3°
Theoretical resolution (azimuth \times ground - range \times elevation)	$0.5\text{m} \times 0.5\text{m} \times 0.5\text{m}$

the 2D image at different sections, which agree with the theoretical values shown in Figures 3(b) and 3(c).

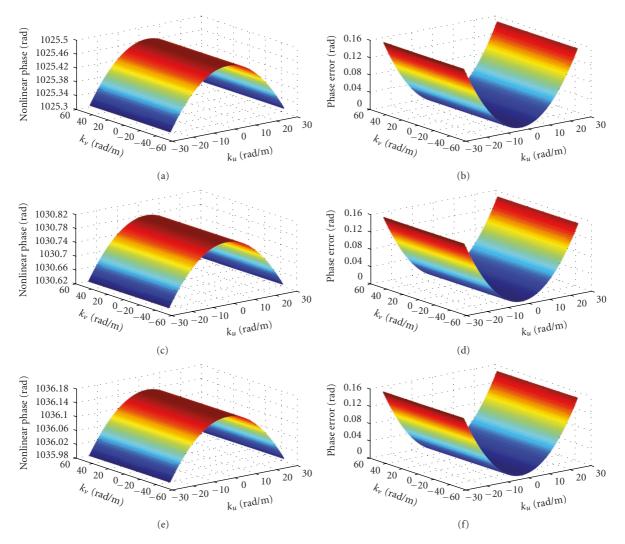


FIGURE 4: The nonlinear phase terms shown in (41) and nonlinear phase error between (41) and (47). (a) nonlinear phase terms calculated at $k = k_{\min}$; (b) nonlinear phase error calculated at $k = k_{\min}$; (c) nonlinear phase terms calculated at $k = k_c$; (d) nonlinear phase error calculated at $k = k_{\max}$; (f) nonlinear phase error calculated at $k = k_{\max}$.

TABLE 2	: Image	quality	parameters.

Point target	Resolution (m)			PSLR (dB)		ISLR (dB)			
	X	Y	Z	X	Y	Ζ	X	Y	Z
А	0.51	0.52	0.53	-13.23	-13.15	-13.18	-10.40	-10.34	-10.33
В	0.52	0.51	0.53	-13.20	-13.17	-13.21	-10.41	-10.20	-10.51
С	0.51	0.51	0.52	-13.24	-13.23	-13.26	-10.25	-10.35	-10.30

Three point targets A, B and C shown in Figure 3(a) are chosen to verify the performance and focusing quality of the proposed 3D RMA, and the image quality parameters, including the resolution, the peak sidelobe ratio (PSLR) and the integrated sidelobe ratio (ISLR), are shown in Table 2. From the above simulation results, we can see that the point targets are all well focused after the removal of constant phase term and compensation of nonlinear phase terms, and the expansion of Φ_0 in (20) is effective without taking the quartic and higher-order terms into account.

5. Influence of Platform Deviations

During the analysis mentioned above, the platform which carries the 3D-SAR system is assumed to fly along a straight line with the constant velocity at the invariable altitude. Our

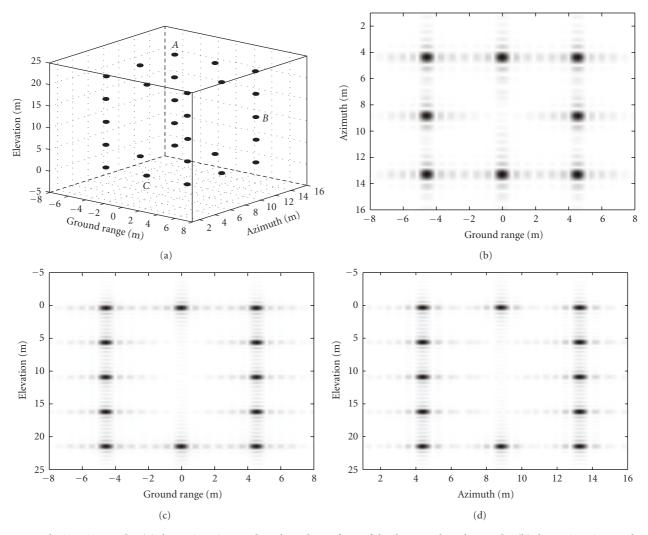


FIGURE 5: The imaging results. (a) the 3D imaging results, where the surfaces of the data are plotted at -3 dB; (b) the 2D imaging results on the 2D *OXY* plane, where Z = 0.5 m; (c) the 2D imaging results on the 2D *OYZ* plane, where X = 4.5 m; (d) the 2D imaging results on the 2D *OXZ* plane, where Y = 4.5 m.

present 3D-SAR with STMR-LAA system is based on the ASTRO, which is a ground-based SAR platform developed at MITL-IECAS. It works in the anechoic chamber and on the ground, and positioning precision as well as control accuracy is high enough to satisfy the above assumption. However, when the system is mounted on the airplane or helicopter platform, trajectory deviations from the straight line, attitude disturbances and forward velocity variations are unavoidable due to atmospheric turbulence. Moreover, the wing oscillation during flying continuously changes the position of every antenna element. These will introduce motion errors to the received raw signal, and may severely impair the final 3D image quality. The impact of platform attitude disturbances on the 3D image quality is presented in Reference [14], and the 3D image quality analysis of the displacement of antenna elements caused by the wing oscillation is given in [12]. In this paper, we analyze the impact of platform trajectory deviations on the 3D image

quality, and present the required accuracy on the platform position to generate the 3D images.

The imaging geometry in the presence of trajectory deviations is shown in Figure 6, where the system axes coincide with the ideal trajectory case shown in Figure 1. The deviation vector **b** equals to $b_x \mathbf{i} + b_y \mathbf{j} + b_z \mathbf{k}$ in the Cartesian coordinates, where b_x , b_y and b_z are the deviation components in the x-axis, y-axis and z-axis, respectively. An analysis on the effect of more complex motion deviation trajectories (pitch-, yaw-, and roll-angle variations) is out of the scope of the paper. The deviation component in the x-axis is caused by the variable velocity which results in unevenly spaced samples in the azimuth direction, and can be compensated via a proper resampling of the received raw signal. Thus it is assumed in the following analysis that the platform trajectory deviation in the x-axis has been compensated and b_x is null. Then the simplified imaging geometry on the zero Doppler plane is shown in Figure 7,

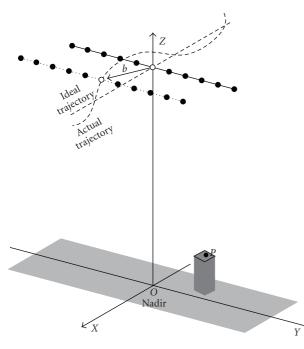


FIGURE 6: Geometry in presence of trajectory deviations.

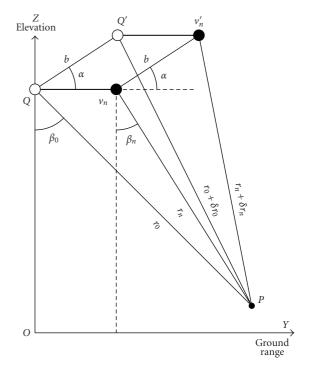


FIGURE 7: Geometry on zero Doppler plane.

where r_0 is the distance from the Tx to the target *P* in the case of ideal trajectory; $r_0 + \delta r_0$ is the distance from the Tx to the target *P* in the case of actual trajectory; r_n is the distance from the target *P* to the *n*th Rx in the case of ideal trajectory; $r_n + \delta r_n$ is the distance from the target *P* to the *n*th Rx in the case of actual trajectory; β_0 denotes the look angle from the Tx to the target *P*. The deviation vector **b** equals to $b(\cos \alpha \mathbf{j} + \sin \alpha \mathbf{k})$ in the polar coordinates.

According to the Cosine Theorem [21], we have

$$r_0 + \delta r_0 = \sqrt{r_0^2 + b^2 - 2r_0 b \sin(\beta_0 - \alpha)}.$$
 (51)

When the value of the deviation is small compared with the slant range of the target, the range error δr_0 in (51) can be approximately expressed as $\delta r_0 \approx b \sin(\alpha - \beta_0)$. Likewise, the range error δr_n can be calculated as $\delta r_n \approx b \sin(\alpha - \beta_n)$. According to (16), the phase error caused by trajectory deviations can be expressed as

$$\delta \Phi = k(\delta r_0 + \delta r_n) \approx 2k_c b \sin\left(\alpha - \frac{\beta_0 + \beta_n}{2}\right) \cos\left(\frac{\beta_0 - \beta_n}{2}\right).$$
(52)

The phase error $\delta \Phi$ varies with the polar angle α when the polar radius *b* is constant. The absolute maximum of $\delta \Phi$ is obtained at $\alpha = \pm \pi/2 + (\beta_0 + \beta_n)/2$, and the absolute minimum, which equals to zero, is obtained at $\alpha = (\beta_0 + \beta_n)/2$ and $\alpha = \pi + (\beta_0 + \beta_n)/2$. Therefore, the whole trajectory deviations can be separate into the deviation component along the equivalent target look direction (i.e., $\alpha = \pm \pi/2 + (\beta_0 + \beta_n)/2$) and the deviation component perpendicular to the equivalent target look direction (i.e., $\alpha = (\beta_0 + \beta_n)/2$ and $\alpha = \pi + (\beta_0 + \beta_n)/2$). The second deviation component contributes nothing to the phase error, and the compensation of platform trajectory deviations only needs to take the first component into account.

The impact of platform trajectory deviations on the 3D image quality can be ignored when the magnitude of phase error caused by trajectory deviations is smaller than $\pi/4$ [22]. According to this principle, the range of required accuracy on the platform position should satisfy the following inequation to generate the 3D image:

$$\left| b_z \cos\left(\frac{\beta_0 + \beta_n}{2}\right) - b_y \sin\left(\frac{\beta_0 + \beta_n}{2}\right) \right| \le \frac{\lambda_c}{16 \cos[(\beta_0 - \beta_n)/2]},$$
(53)

where $b_y = b \cos \alpha$ and $b_z = b \sin \alpha$. It can be calculated with the parameters given in Table 1 that the platform position should be known with accuracy up to the magnitude of millimeter. For platform deviations larger than that, the PSLR and ISLR will be increased drastically. It demonstrates that the platform position has to be measured or estimated very precisely in order to form high quality 3D images.

6. Conclusions

This paper presents a 3D RMA suitable for downwardlooking 3D-SAR with STMR-LAA. The precise implicit expression of the signal spectrum in 3D wavenumber domain is firstly described. Then the approximate explicit expression of the signal spectrum is derived by expanding the implicit phase term to its Taylor series. After the removal of constant phase term and compensation of nonlinear phase terms, a 3D wavenumber mapping is proposed to replace the wavenumber in the direction of wave propagation by the wavenumber observed in 3D spatial domain *OXYZ*, and an interpolation is performed to transform these unevenly spaced samples into evenly spaced ones in 3D wavenumber domain. The simulation results about the imaging quality and the phase error analysis validate the feasibility of the imaging algorithm. Furthermore, the impact of platform trajectory deviations on the 3D image quality is discussed, and it concludes that the platform position has to be known precisely to form high quality 3D images.

Future work will extend the proposed 3D RMA to more complex bistatic configurations of downward-looking 3D-SAR with LAA, such as multi-transmitting and multireceiving case. In addition, this paper contains neither technological aspects about the realization of airborne 3D-SAR with LAA system nor the key techniques, including the motion compensation, wings oscillation compensation and relevant problems. These special problems have to be treated separately when the 3D-SAR with LAA system will be mounted on the airplane or helicopter platform in the future.

Appendices

A. Derivation of the First Partial Derivatives of Φ_0 with Respect To y_T and z_T

According to (16), the first partial derivative of Φ_0 with respect to y_T can be written as

$$\frac{\partial \Phi_0}{\partial y_T} = k \frac{\partial R(U_0, V_0)}{\partial y_T} + k_u \frac{\partial U_0}{\partial y_T} + k_v \frac{\partial V_0}{\partial y_T}, \quad (A.1)$$

where

$$R(U_0, V_0) = R_T(U_0) + R_R(U_0, V_0),$$
(A.2)

$$R_T(U_0) = \sqrt{U_0^2 + y_T^2 + (H - z_T)^2},$$
 (A.3)

$$R_R(U_0, V_0) = \sqrt{U_0^2 + V_0^2 + (H - z_T)^2}.$$
 (A.4)

We can see from (A.2) that $R(U_0, V_0)$ are the function of U_0 , V_0 , y_T and z_T , and the stationary phase points U_0 and V_0 are the function of y_T and z_T . Therefore,

$$\begin{aligned} \frac{\partial R(U_0, V_0)}{\partial y_T} &= R_{y_T}(U_0, V_0) + R_{U_0}(U_0, V_0) \frac{\partial U_0}{\partial y_T} \\ &+ R_{V_0}(U_0, V_0) \frac{\partial V_0}{\partial y_T} \\ &= \frac{y_T}{R_T(U_0)} + \left[\frac{U_0}{R_T(U_0)} + \frac{U_0}{R_R(U_0, V_0)} \right] \frac{\partial U_0}{\partial y_T} \\ &+ \frac{V_0}{R_R(U_0, V_0)} \frac{\partial V_0}{\partial y_T}, \end{aligned}$$
(A.5)

where $R_{y_T}(U_0, V_0)$, $R_{U_0}(U_0, V_0)$ and $R_{V_0}(U_0, V_0)$ denote the first derivatives of $R(U_0, V_0)$ with respect to y_T , U_0 and V_0 . Inserting (A.5) into (A.1) yields

$$\begin{aligned} \frac{\partial \Phi_0}{\partial y_T} &= k \frac{y_T}{\sqrt{U_0^2 + y_T^2 + (H - z_T)^2}} \\ &+ \left[k \frac{V_0}{\sqrt{U_0^2 + V_0^2 + (H - z_T)^2}} + k_\nu \right] \frac{\partial V_0}{\partial y_T} \\ &+ \left[k \frac{U_0}{\sqrt{U_0^2 + y_T^2 + (H - z_T)^2}} \right] \\ &+ k \frac{U_0}{\sqrt{U_0^2 + V_0^2 + (H - z_T)^2}} + k_\mu \frac{\partial U_0}{\partial y_T}. \end{aligned}$$
(A.6)

According to equations set (14), the terms in the two square brackets of (A.6) are equal to zero. Therefore, (A.6) can be simplified to be

$$\frac{\partial \Phi_0}{\partial y_T} = k \frac{y_T}{\sqrt{U_0^2 + y_T^2 + (H - z_T)^2}}.$$
 (A.7)

Likewise, the first partial derivative of Φ_0 with respect to z_T can be calculated.

$$\begin{split} \frac{\partial \Phi_{0}}{\partial z_{T}} \\ &= k \frac{\partial R(U_{0}, V_{0})}{\partial z_{T}} + k_{u} \frac{\partial U_{0}}{\partial z_{T}} + k_{v} \frac{\partial V_{0}}{\partial z_{T}} \\ &= k \left\{ \frac{z_{T} - H}{\sqrt{U_{0}^{2} + y_{T}^{2} + (H - z_{T})^{2}}} + \frac{z_{T} - H}{\sqrt{U_{0}^{2} + V_{0}^{2} + (H - z_{T})^{2}}} \right\} \\ &+ \left[k \frac{V_{0}}{\sqrt{U_{0}^{2} + V_{0}^{2} + (H - z_{T})^{2}}} + k_{v} \right] \frac{\partial V_{0}}{\partial z_{T}} \\ &+ \left[k \frac{U_{0}}{\sqrt{U_{0}^{2} + y_{T}^{2} + (H - z_{T})^{2}}} + k_{u} \right] \frac{\partial U_{0}}{\partial z_{T}} \\ &+ k \frac{U_{0}}{\sqrt{U_{0}^{2} + V_{0}^{2} + (H - z_{T})^{2}}} + k_{u} \right] \frac{\partial U_{0}}{\partial z_{T}} \\ &= \frac{z_{T} - H}{U_{0}} \left\{ k \frac{U_{0}}{\sqrt{U_{0}^{2} + y_{T}^{2} + (H - z_{T})^{2}}} + k \frac{U_{0}}{\sqrt{U_{0}^{2} + y_{T}^{2} + (H - z_{T})^{2}}} \right\} \\ &= k_{u} \frac{H - z_{T}}{U_{0}}. \end{split}$$
(A.8)

B. Derivation of the First Partial Derivatives of U_0 and V_0 with Respect to y_T and z_T Evaluated at $y_T = 0$, $z_T = 0$

The first partial derivatives of equations set (14) with respect to y_T can be written as

$$k\frac{\partial}{\partial y_{T}}\left[\frac{U_{0}}{\sqrt{U_{0}^{2}+y_{T}^{2}+(H-z_{T})^{2}}}+\frac{U_{0}}{\sqrt{U_{0}^{2}+V_{0}^{2}+(H-z_{T})^{2}}}\right]=0,$$

$$k\frac{\partial}{\partial y_{T}}\left[\frac{V_{0}}{\sqrt{U_{0}^{2}+V_{0}^{2}+(H-z_{T})^{2}}}\right]=0.$$
(B.1)

Expanding (B.1), we have

$$-\frac{y_T U_0}{R_T^3(U_0)} + \left[\frac{y_T^2 + (H - z_T)^2}{R_T^3(U_0)} + \frac{V_0^2 + (H - z_T)^2}{R_R^3(U_0, V_0)}\right] \frac{\partial U_0}{\partial y_T} - \frac{U_0 V_0}{R_R^3(U_0, V_0)} \frac{\partial V_0}{\partial y_T} = 0, - \frac{U_0 V_0}{R_R^3(U_0, V_0)} \frac{\partial U_0}{\partial y_T} + \frac{U_0^2 + (H - z_T)^2}{R_R^3(U_0, V_0)} \frac{\partial V_0}{\partial y_T} = 0.$$
(B.2)

Therefore, the first partial derivates of U_0 and V_0 with respect to y_T evaluated at $y_T = 0$, $z_T = 0$ can be calculated as follows:

$$\left[\frac{H^2}{\left(U_{00}^2 + H^2\right)^{3/2}} + \frac{V_{00}^2 + H^2}{\left(U_{00}^2 + V_{00}^2 + H^2\right)^{3/2}}\right] \frac{\partial U_0}{\partial y_T}\Big|_{y_T=0, z_T=0} - \frac{U_{00}V_{00}}{\left(U_{00}^2 + V_{00}^2 + H^2\right)^{3/2}} \frac{\partial V_0}{\partial y_T}\Big|_{y_T=0, z_T=0} = 0,$$

$$- U_{00}V_{00} \frac{\partial U_0}{\partial y_T}\Big|_{y_T=0, z_T=0} + \left(U_{00}^2 + H^2\right) \frac{\partial V_0}{\partial y_T}\Big|_{y_T=0, z_T=0} = 0.$$
(B.3)

For the determinant of coefficient matrix in equations set (B.3) is unequal to zero, that is,

$$\begin{vmatrix} \frac{H^2}{\left(U_{00}^2 + H^2\right)^{3/2}} + \frac{V_{00}^2 + H^2}{\left(U_{00}^2 + V_{00}^2 + H^2\right)^{3/2}} & \frac{-U_{00}V_{00}}{\left(U_{00}^2 + V_{00}^2 + H^2\right)^{3/2}} \\ -U_{00}V_{00} & U_{00}^2 + H^2 \end{vmatrix}$$

$$= \frac{H^2}{\sqrt{U_{00}^2 + H^2}} + \frac{H^2}{\sqrt{U_{00}^2 + V_{00}^2 + H^2}} \neq 0,$$
(B.4)

the solution of equations set (B.3) is

$$\frac{\partial U_0}{\partial y_T} \bigg|_{y_T=0, z_T=0} = 0,$$

$$\frac{\partial V_0}{\partial y_T} \bigg|_{y_T=0, z_T=0} = 0.$$
(B.5)

Likewise, the first partial derivatives of equations set (14) with respect to z_T can be written as

$$k\frac{\partial}{\partial z_{T}}\left[\frac{U_{0}}{\sqrt{U_{0}^{2}+y_{T}^{2}+(H-z_{T})^{2}}}+\frac{U_{0}}{\sqrt{U_{0}^{2}+V_{0}^{2}+(H-z_{T})^{2}}}\right]=0,$$

$$k\frac{\partial}{\partial z_{T}}\left[\frac{V_{0}}{\sqrt{U_{0}^{2}+V_{0}^{2}+(H-z_{T})^{2}}}\right]=0.$$
(B.6)

Expanding (B.6), we have

$$\frac{U_0(H-z_T)}{R_T^3(U_0)} + \frac{U_0(H-z_T)}{R_R^3(U_0,V_0)} \\
+ \left[\frac{y_T^2 + (H-z_T)^2}{R_T^3(U_0)} + \frac{V_0^2 + (H-z_T)^2}{R_R^3(U_0,V_0)}\right] \frac{\partial U_0}{\partial z_T} \\
- \frac{U_0 V_0}{R_R^3(U_0,V_0)} \frac{\partial V_0}{\partial z_T} = 0,$$
(B.7)

$$egin{aligned} &rac{V_0(H-z_T)}{R_R^3(U_0,V_0)} - rac{U_0V_0}{R_R^3(U_0,V_0)} rac{\partial U_0}{\partial z_T} \ &+ rac{U_0^2 + (H-z_T)^2}{R_R^3(U_0,V_0)} rac{\partial V_0}{\partial z_T} = 0. \end{aligned}$$

Therefore, the first partial derivates of U_0 and V_0 with respect to z_T evaluated at $y_T = 0$, $z_T = 0$ can be calculated as follows:

$$\begin{split} \frac{U_{00}H}{\left(U_{00}^{2}+H^{2}\right)^{3/2}} &+ \frac{U_{00}H}{\left(U_{00}^{2}+V_{00}^{2}+H^{2}\right)^{3/2}} \\ &+ \left[\frac{H^{2}}{\left(U_{00}^{2}+H^{2}\right)^{3/2}} + \frac{V_{00}^{2}+H^{2}}{\left(U_{00}^{2}+V_{00}^{2}+H^{2}\right)^{3/2}}\right] \frac{\partial U_{0}}{\partial z_{T}}\Big|_{y_{T}=0, z_{T}=0} \\ &- \frac{U_{00}V_{00}}{\left(U_{00}^{2}+V_{00}^{2}+H^{2}\right)^{3/2}} \frac{\partial V_{0}}{\partial z_{T}}\Big|_{y_{T}=0, z_{T}=0} = 0, \\ V_{00}H - U_{00}V_{00} \frac{\partial U_{0}}{\partial z_{T}}\Big|_{y_{T}=0, z_{T}=0} \end{split}$$

$$+ \left(U_{00}^{2} + H^{2}\right) \frac{\partial V_{0}}{\partial z_{T}} \Big|_{y_{T}=0, z_{T}=0} = 0.$$
(B.8)

Solving the equations set (B.8), we obtain

$$\left. \frac{\partial U_0}{\partial z_T} \right|_{y_T=0, z_T=0} = -\frac{U_{00}}{H},$$

$$\left. \frac{\partial V_0}{\partial z_T} \right|_{y_T=0, z_T=0} = -\frac{V_{00}}{H}.$$
(B.9)

C. Derivation of the Second Partial Derivatives of U_0 and V_0 with Respect to z_T Evaluated at $y_T = 0, z_T = 0$

The first partial derivatives of equations set (B.7) with respect to z_T can be written as

$$\begin{aligned} \frac{\partial}{\partial z_T} \left\{ \frac{U_0(H-z_T)}{R_T^3(U_0)} + \frac{U_0(H-z_T)}{R_R^3(U_0,V_0)} \\ &+ \left[\frac{y_T^2 + (H-z_T)^2}{R_T^3(U_0)} + \frac{V_0^2 + (H-z_T)^2}{R_R^3(U_0,V_0)} \right] \frac{\partial U_0}{\partial z_T} \\ &- \frac{U_0 V_0}{R_R^3(U_0,V_0)} \frac{\partial V_0}{\partial z_T} \right\} = 0, \end{aligned}$$
(C.1)

$$egin{aligned} &rac{\partial}{\partial z_T} \left\{ rac{V_0(H-z_T)}{R_R^3(U_0,V_0)} - rac{U_0V_0}{R_R^3(U_0,V_0)} rac{\partial U_0}{\partial z_T}
ight. \ &+ rac{U_0^2 + (H-z_T)^2}{R_R^3(U_0,V_0)} rac{\partial V_0}{\partial z_T}
ight\} = 0. \end{aligned}$$

Expanding (C.1) and inserting (B.9) into it, we can obtain that the second partial derivatives of U_0 and V_0 with respect to z_T evaluated at $y_T = 0$, $z_T = 0$ satisfy the following equations set, that is,

$$\left[\frac{H^2}{\left(U_{00}^2 + H^2\right)^{3/2}} + \frac{V_{00}^2 + H^2}{\left(U_{00}^2 + V_{00}^2 + H^2\right)^{3/2}}\right] \frac{\partial^2 U_0}{\partial z_T^2}\Big|_{y_T=0, z_T=0} - \frac{U_{00}V_{00}}{\left(U_{00}^2 + V_{00}^2 + H^2\right)^{3/2}} \frac{\partial^2 V_0}{\partial z_T^2}\Big|_{y_T=0, z_T=0} = 0,$$

$$- U_{00}V_{00} \frac{\partial^2 U_0}{\partial z_T^2}\Big|_{y_T=0, z_T=0} + \left(U_{00}^2 + H^2\right) \frac{\partial^2 V_0}{\partial z_T^2}\Big|_{y_T=0, z_T=0} = 0.$$
 (C.2)

For equations set (C.2) have the same coefficient matrix with equations set (B.3), we can obtain the solution of equations set (C.2) as follows.

$$\frac{\partial^2 U_0}{\partial z_T^2} \bigg|_{y_T=0, \ z_T=0} = 0,$$

$$\frac{\partial^2 V_0}{\partial z_T^2} \bigg|_{y_T=0, \ z_T=0} = 0.$$
(C.3)

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